

Unidad de Investigación y Desarrollo

G.E.M.A.

Grupo de Ensayos Mecánicos Aplicados



RESEARCH AND DEVELOPMENT ACTIVITIES

Ing. Pablo L. Ringegni



FACULTAD DE INGENIERÍA

EDUCACIÓN
PÚBLICA
Y GRATUITA



UNIVERSIDAD
NACIONAL
DE LA PLATA

CTA STRUCTURE

UNIVERSIDAD NACIONAL
DE LA PLATA



Departamento de Aeronáutica

UID G.E.M.A.

UID GFC

UID LaClyFa





CTA Structure

Unidad de Investigación y Desarrollo

G.E.M.A.

Grupo de Ensayos Mecánicos Aplicados

Applied Mechanical Testing Group (GEMA)
Their work and research is related with **structures, mechanics and thermal**. Specially related with numerical simulation (finite elements) and testing. Its activities are within the area of dynamic loads (fatigue and vibration), analysis and simulation of structures and mechanisms, thermal behavior, space technology, design, testing and integration, and many other related activities. It has also obtained ISO 17025 accreditation for some of their tests and its quality system is certified under ISO 9001.



Laboratory of Boundary Layer and Environmental Fluidynamics (LaCLyFA)

Their works interdisciplinary with engineers, physicists, architects, meteorologists and biologists **in problematic associated with wind engineering and aerodynamics**.



Computational Fluid Dynamic Group (GFC)
Their work is related to the **numerical analysis and simulation of fluid dynamic and aerodynamics problems**.

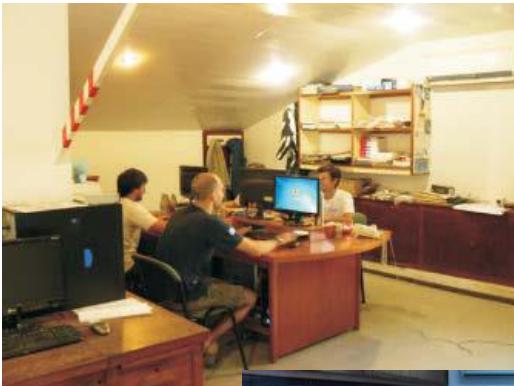


UID G.E.M.A.

Research and Development Unit

Applied Mechanics Testing Group

(Since 1994



WORKING TEAM (50 people)

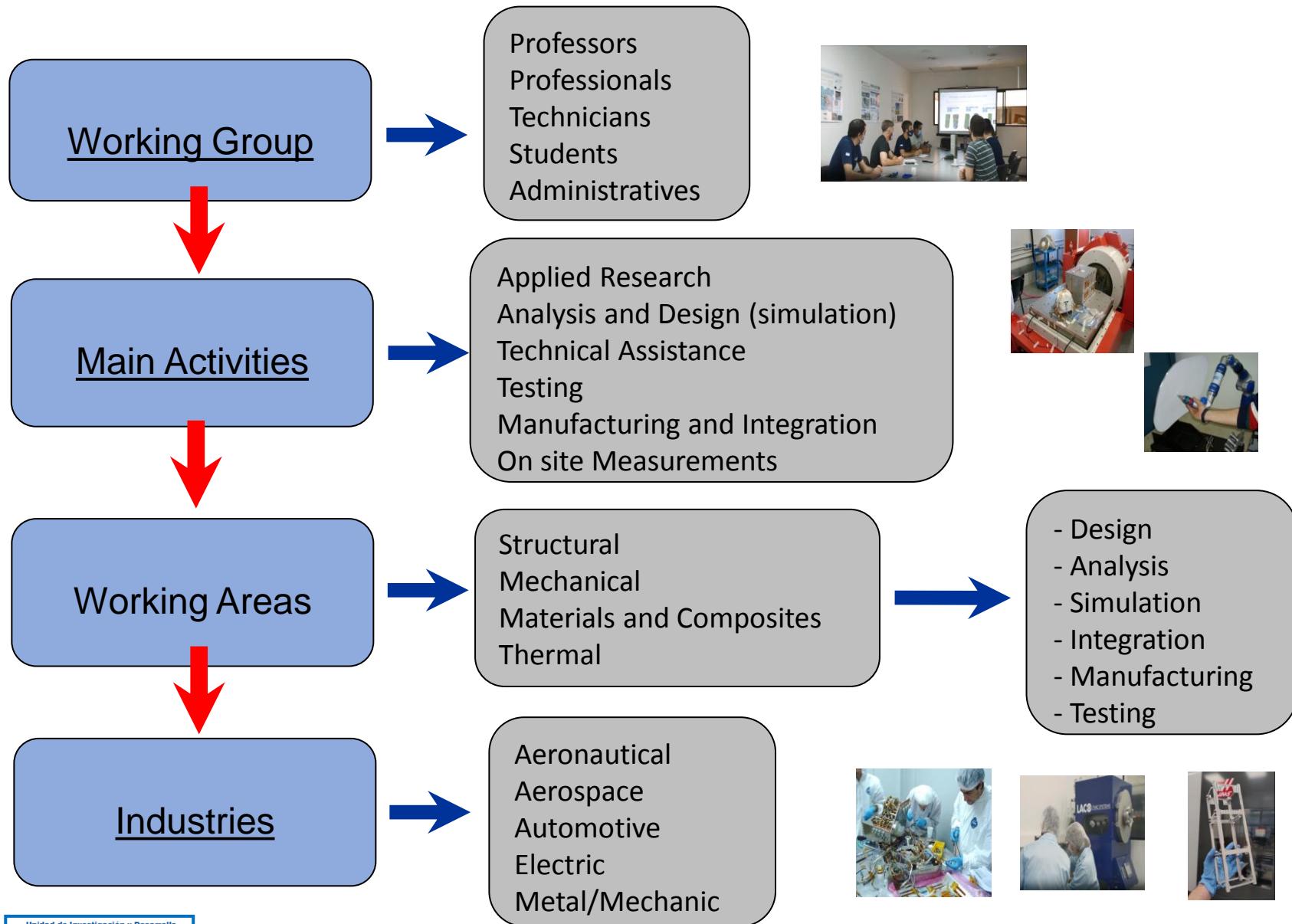
- Professors: Doctors, Masters, Specialists, Engineers
- Professionals
- Technicians
- Students (Scholarship holders)

TYPE OF ACTIVITIES

- Applied Research
- Technical Development and Assistance
- Design, Analysis, Integration and Testing

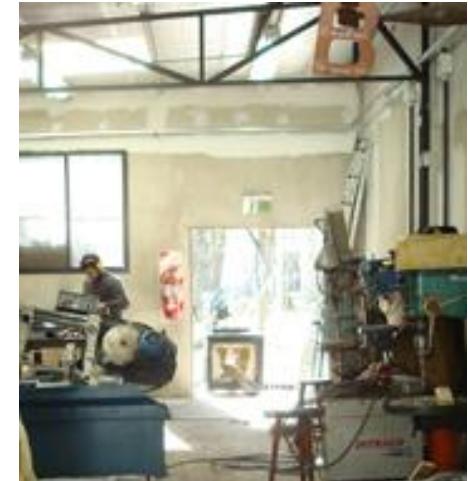


UID GEMA STRUCTURE



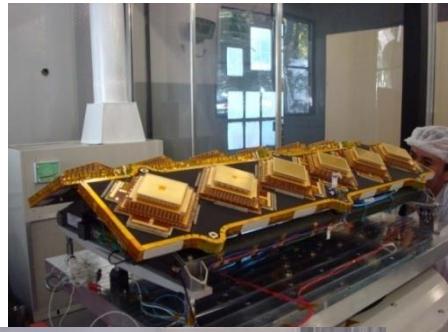
INFRASTRUCTURE

Testing Laboratories: (2) (800 y 270 m2)

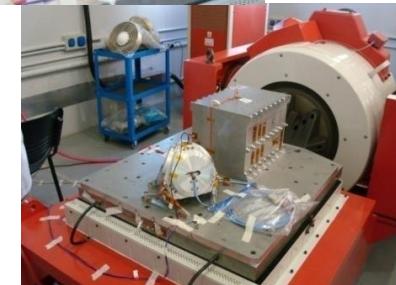


INFRASTRUCTURE

Structural and Thermal TESTS of Aerospace Componentes



Mechanical



Thermal



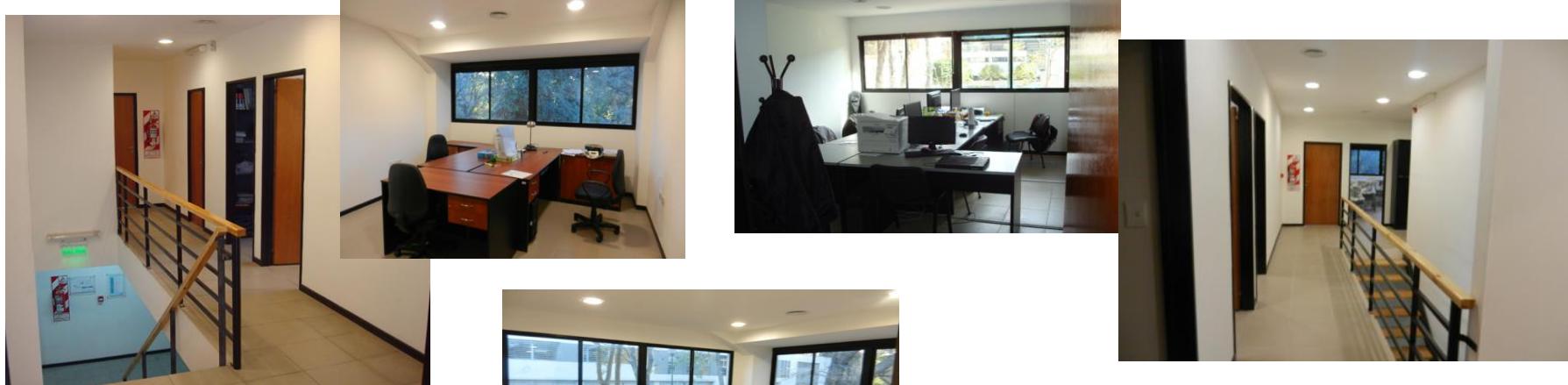
INFRASTRUCTURE

Two (2) Clean Room / Integration (26 m² and 30 m²)



INFRAESTRUCTURA

Offices (Simulation, Análisys and Design): 350 m2



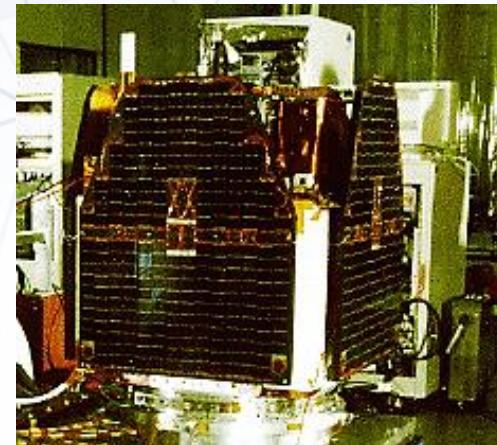
Meeting Room (2)
(16 m2 /10 p- 45 m2 /25p)



Past Aerospace Projects and Activities

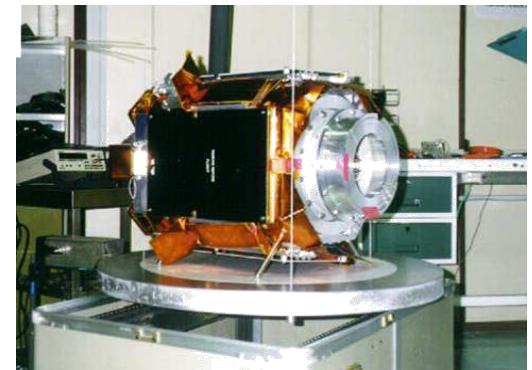
SCIENTIFIC ARGENTINE SATELLITE: SAC-B (CONAE) - 1994

Mechanical Tests and Mass Properties Determination (PF Units and Development Units)



SCIENTIFIC ARGENTINE SATELLITE: SAC-A (CONAE) - 1997

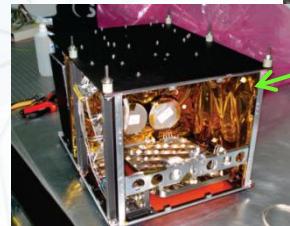
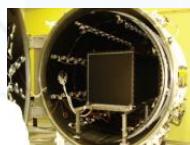
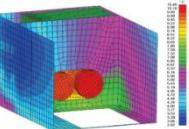
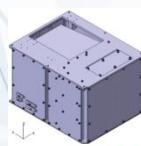
Mass Properties Determination



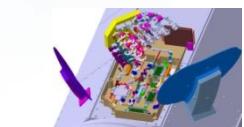
Past Aerospace Projects and Activities

SCIENTIFIC ARGENTINE SATELLITE SAC-D (CONAE) SAC-D/Aquarius (NASA) MISSION - 2006-2011

**NIRST
Instrument**



**MWR
Instrument**



Four of five Argentina's Instruments were developed at the UNLP laboratories with the participation of IAR, CIOP, LEICI, CETAD and GEMA

Activities for MWR and NIRST

- Structural, Mechanical and Thermal Design, Simulation and Analysis.
- Structure and Thermal Hardware manufacture.
- PF Instruments Integration
- Environmental Tests: Structural (Vibrations and Shock) and Thermal.
- Mass Properties: CG., Mass, M.I.

Activities for different SAC D's Instruments:

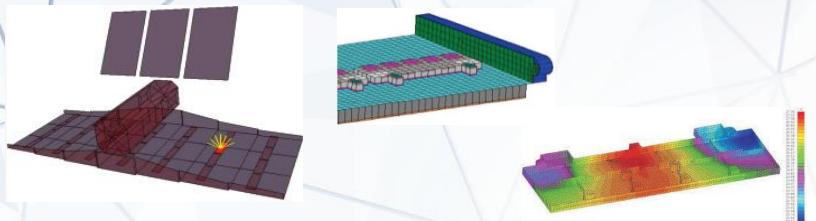
- Mechanical Vibration Tests (ROA, ROSA, GPS, PAD, TDP, X and S Antennas)
- MLI design and manufacture (SP, S and X band Antennas)



Past Aerospace Projects and Activities

SCIENTIFIC ARGENTINE SATELLITE SAOCOM 1A - 1B (CONAE) 2011 - 2020

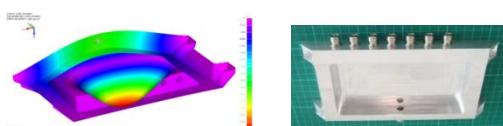
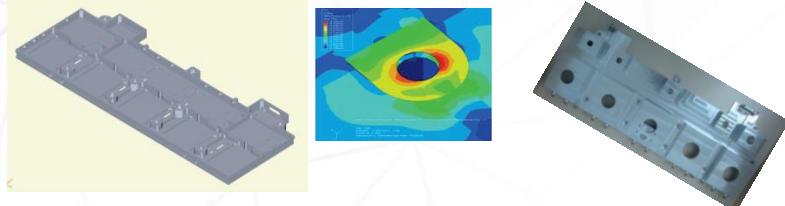
Thermal analysis and design of SAOCOM SAR Antenna and its components



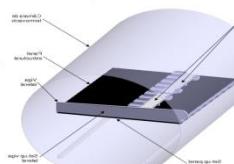
SAOCOM SAR
Antenna



SAOCOM SAR Antenna components structural analysis and design



Design and performing thermo-vacuum and structural tests of SAOCOM SAR Antenna components



Design, construction and integration of different devices and thermal hardware (MLI).



Past Aerospace Projects and Activities

SCIENTIFIC ARGENTINE SATELLITE SAOCOM 1A - 1B (CONAE)

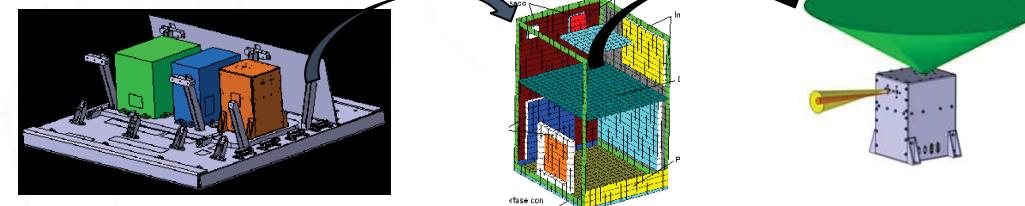
2011 - 2020

Structural and Thermal Analysis of SAOCOM S Band and X band Antennas and Tests



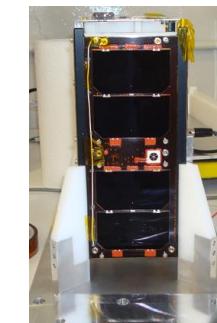
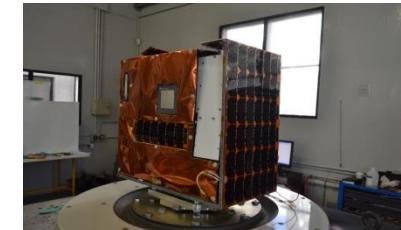
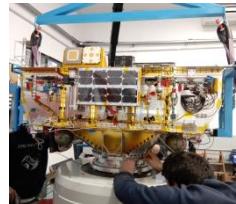
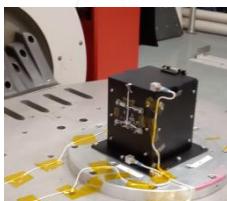
ARSAT 1 (ARSAT - INVAP CNEA) - 2010

Structural and Thermal Design and Analysis of MARE (Space Radiation Argentine Monitor)



SATELLOGIC / Innova Space / Immer / EPIC- 2010-2023

Performing Structural Tests over Small Satellites and Platforms

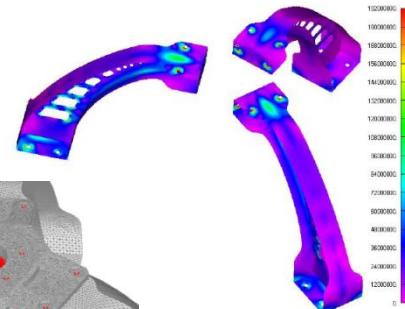
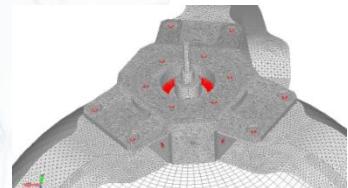
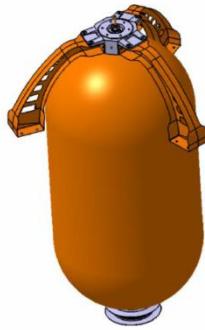


Past Aerospace Projects and Activities

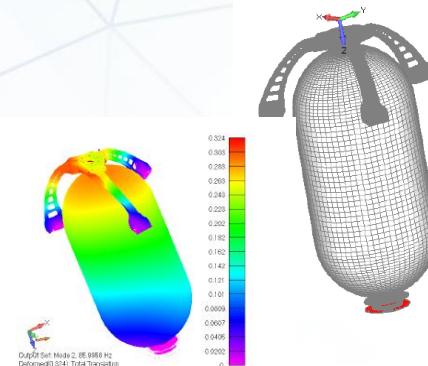
INVAP ARSAT 1 & ARSAT 2 - 2010

Structural Analysis of Satellite's Components

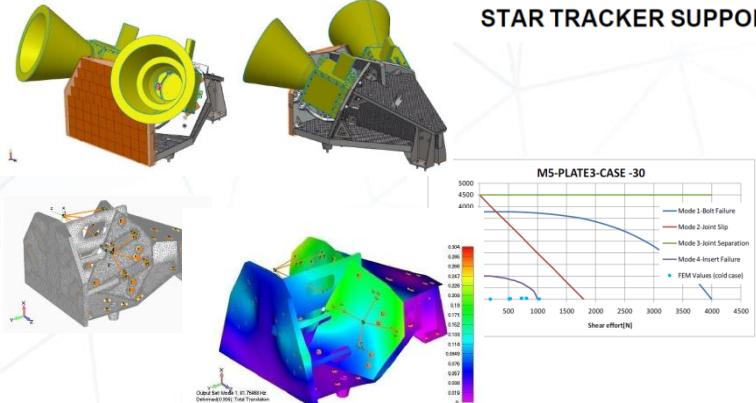
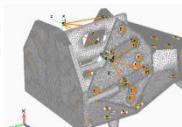
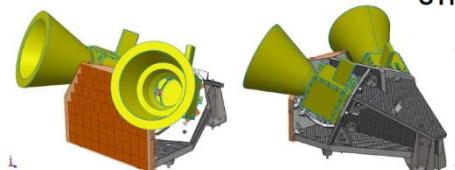
ARSAT 1 HELIUM TANK SUPPORT



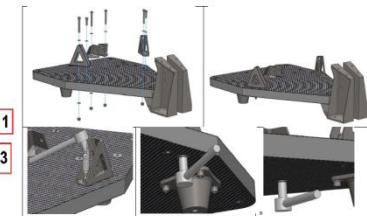
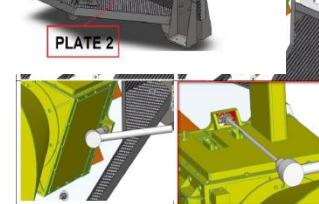
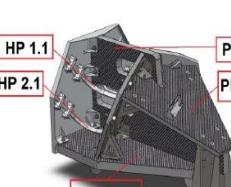
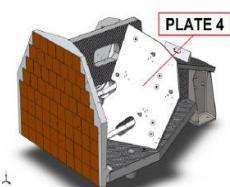
STRUCTURAL ANALYSIS OF ARSAT 2 HE TANK SUPPORT SIMPLIFIED FE MODEL



STRUCTURAL ANALYSIS OF ARSAT 2 STAR TRACKER SUPPORT

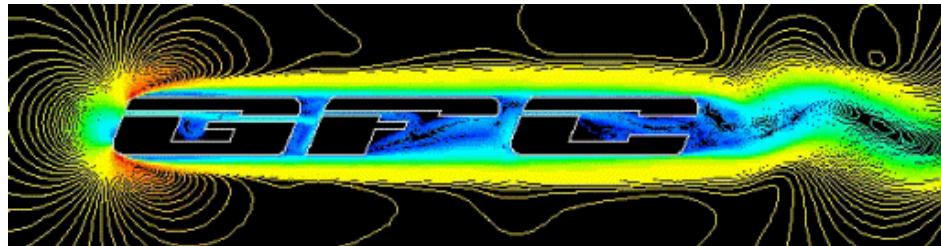


STRUCTURAL DESIGN OF ARSAT 2 STAR TRACKER SUPPORT



UID Computational Fluid Dynamic Group (GFC)

Directors : Professor. Ana Scarabino / Professor Federico Bacchi



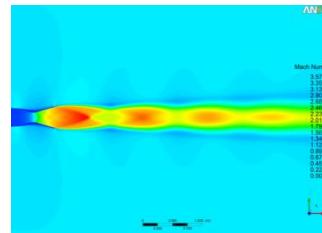
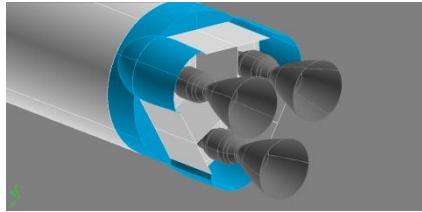
Numerical Analysis and Simulation of Fluid Dynamic and Aerodynamics Problems.

MAIN GOALS

- Development and Validation of CFD codes for Research and Analysis of complex problems.
- Numerical and Experimental Research in Fluid Mechanics.
- Fluid Dynamic and Aerodynamic research Activities (Thermodynamics and Propulsion).
- HR Training and Postgraduate Courses on Fluid Mechanics, Aerodynamics and Turbomachinery.

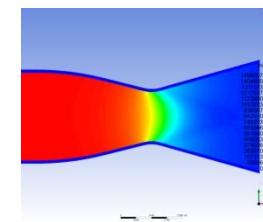
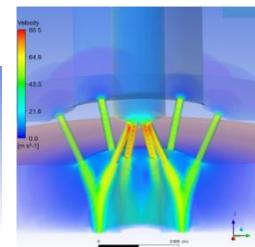
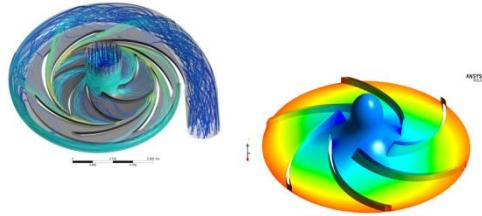
Activities - PROJECTS

Deflector jet Design of a rocket engine



Fluid Dynamic and Thermal Analysis of rocket engine

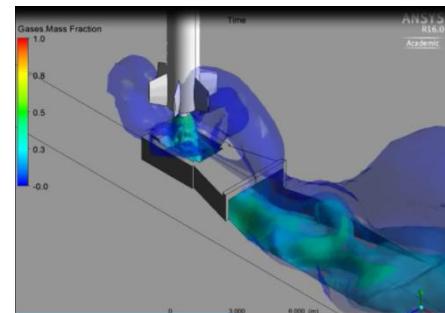
Rocket motor bipropellant injectors Design



Nozzle Pressure Analysis

Turbopump Flow Simulation and Analysis

Sloshing Analysis in fuel tank



Jet Rocket Analysis on ramp

UID LaCLyFA

[Research And Development Unit – Boundary Layer and Environmental Fluid Dynamic Laboratory]

Directors : Professor Julio Marañón / Sebastián Delnero

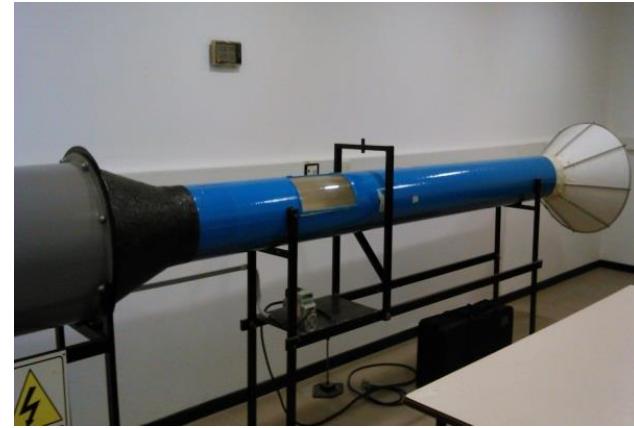


Fluid Dynamic and Aerodynamics Design and Tests.



UID LaCLyFA

[Research And Development Unit – Boundary Layer and Environmental Fluid Dynamic Laboratory]



Test Laboratory: Wind Tunels



PROJETS – LAUNCH VEHICLES

TRONADOR II LAUNCH VEHICLE PROJECT

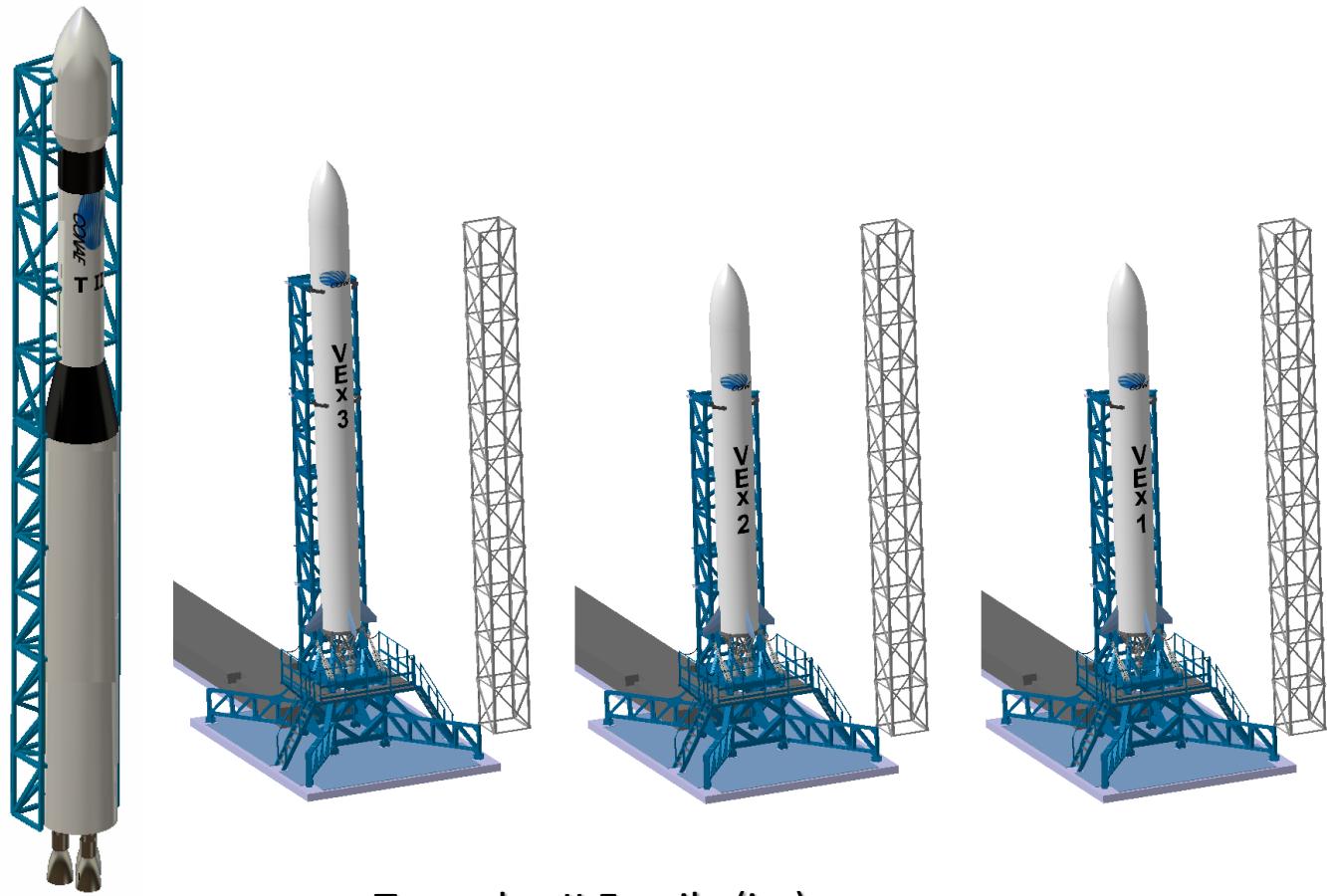


Conceptual and Preliminary Design for the integral project to develop structures, mechanical, auxiliar systems and mechanical GSE of the TRONADOR II prototype.

GOALS:

- Develop the scientific and technical knowledge for the conceptual design to reach a better comprehension of structural and thermal behaviour of elements, components and aerospace structures.
- Design engineering prototypes and monitor manufacturing process.
- Foresee the manufacture of minimum 6 prototypes or experimental vehicles towards to reach the final vehicle (Tronador II)

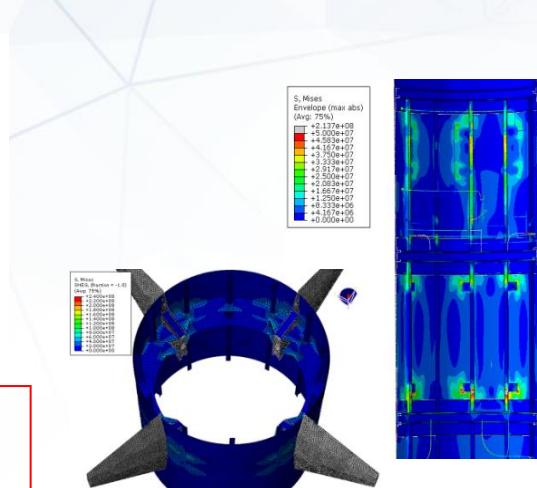
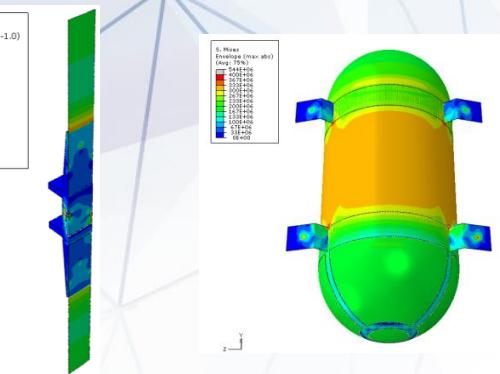
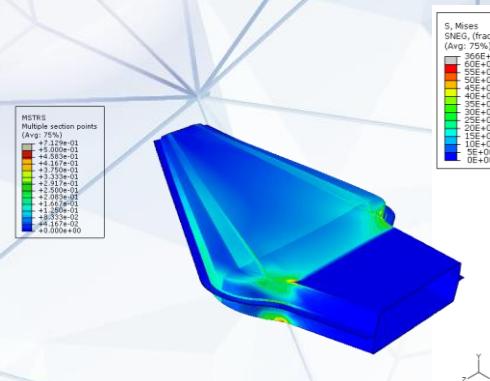
PROJETS – LAUNCH VEHICLES



Tronador II Family (i.e)

Past Aerospace Projects and Activities

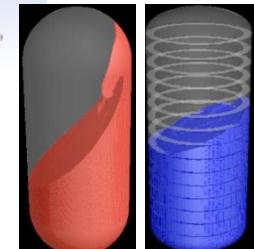
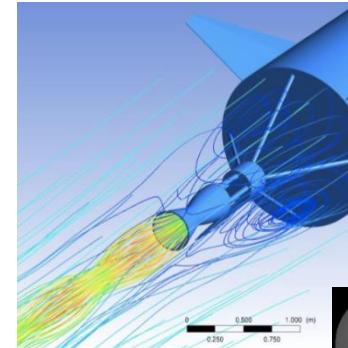
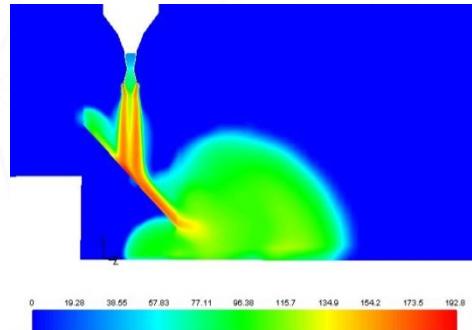
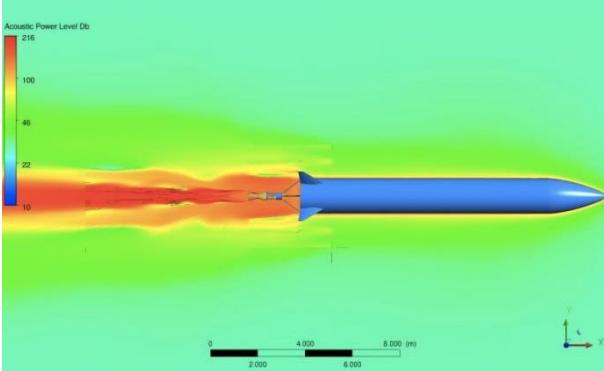
VEx1



Structures

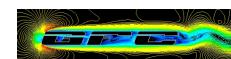
Unidad de Investigación y Desarrollo
G.E.M.A.
Grupo de Ensayos Mecánicos Aplicados

Analysis and Simulation (statics and dynamics) / Design / Testing



Fluid Mechanics

Analysis / Simulation / Performance analysis
/ Pogo / Sloshing / Antivortex / Piping



Past Aerospace Projects and Activities



VEx1

VEx 1 in La Plata Laboratory

Integration and Manufacturing



Man Power



Past Aerospace Projects and Activities



VEx1 at Launch Pad



VEx1



Past Aerospace Projects and Activities



VEx5A

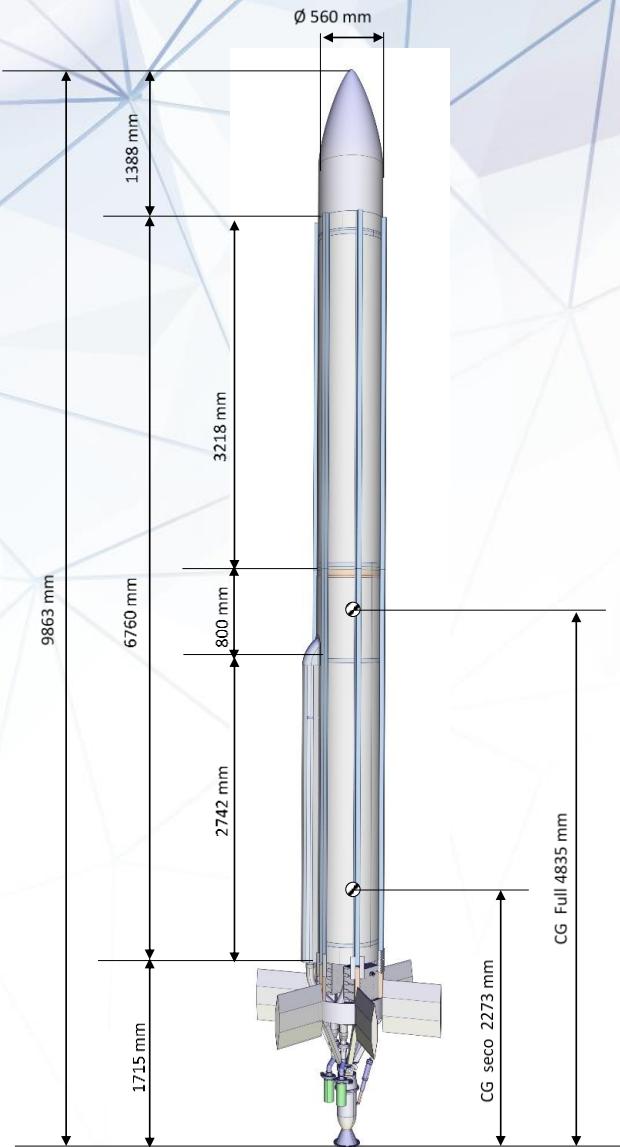


Design / Layout
Manufacturing
Structural Analysis

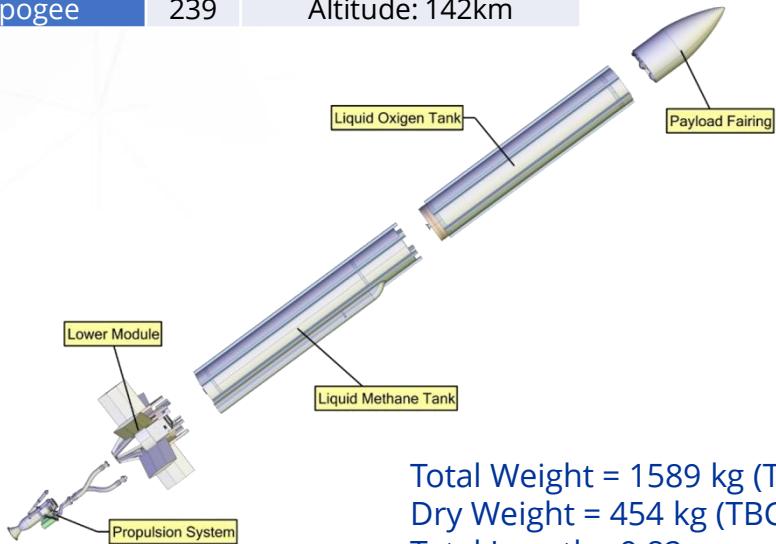


Today and Future Projects

VE-CTA Performance



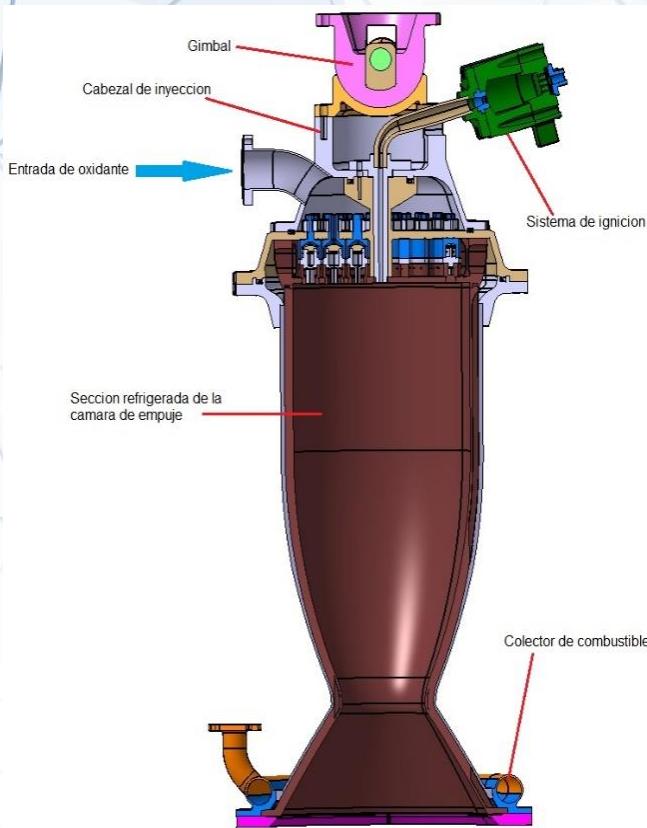
Events	t (s)	Description
Fire	-2	Thrust (nominal at SL 25kN)
Lift Off	0	Accel: 1,53g
Mach 1	44	Altitude: 6,7km
Max Q	50	30kPa, Altitude: 8,9km
Max accel.	119	5,8g, Altitude: 69,2km
MECO	120	Altitude: 70,7km
Fairing Opening	133	Altitude: 85km
Apogee	239	Altitude: 142km



Total Weight = 1589 kg (TBC)
 Dry Weight = 454 kg (TBC)
 Total Length = 9.82 m
 Aluminum Tanks
 CRP Fins and Fairing
 LiPo/LiFePO4 Batteries

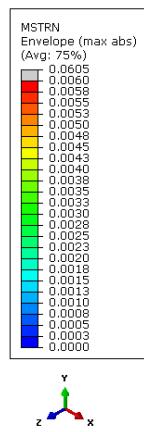
Today and Future Projects

VE-CTA Rocket Engine – Electric pump FED

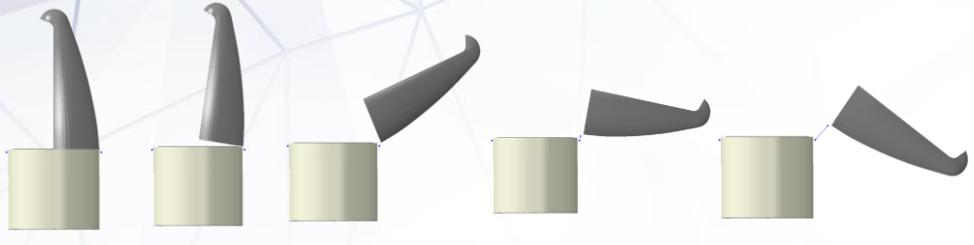


Fuel	Liquid Methane (LCH ₄)	--
Oxidant	Liquid Oxygen (LOX)	--
Feed System	Electropumps	
Absolute pressure inside combustion chamber	35	Kg/cm ²
Adapted Nozzle Exit Plane Absolute pressure (P_e)	0,01 (0,00101)	kg/cm ² (MPa)
Mix ratio ($R = O/F$)	3,2	--
Thrust (adapted Nozzle)	3500	Kgf
Thrust (vacuum)	3602	kgf
Design altitude of nozzle (Adapted pressure)	31150 (0,01)	m (kg/cm ²)
Expansion ratio (optimum) (ε_e)	215,1	--
Burning Time (nominal)	400	Sec
Thrust chamber design type	Cooled until Patm + radiative	--
Thrust control	Throttling de TBD %	--
Injectors	Coaxial – swirl	--
Specific Impulse (I_{VAC})	391,5	seg
Total Mass flow (gases) (\dot{m}_M)	9,44	Kg/s
Fuel Total Mass Flow (\dot{m}_{M-f})	2,24	Kg/s
Fuel Average Density (ρ_f)	429,37	Kg/m ³
Fuel Total Volumetric Flow (Q_{M-f})	5,23 (0,00523)	litros (m ³ /seg)
Oxidant Total Mass Flow (\dot{m}_{M-ox})	7,19	Kg/s
Oxidant Average Density (ρ_{ox})	1151,15	Kg/m ³
Oxidant Total Volumetric Flow (Q_{M-ox})	6,25 (0,00625)	liters (m ³ /sec)
Thrust at sea level (F_{SL})	2595,67 (25,43)	kgf (kN)

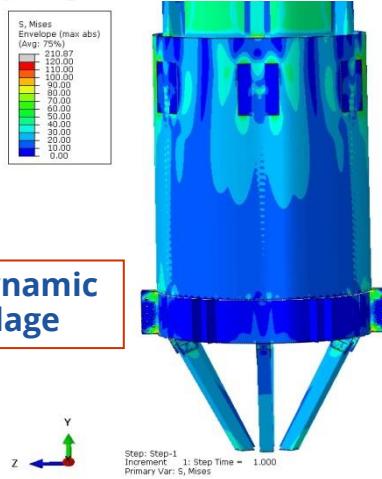
Today and Future Projects



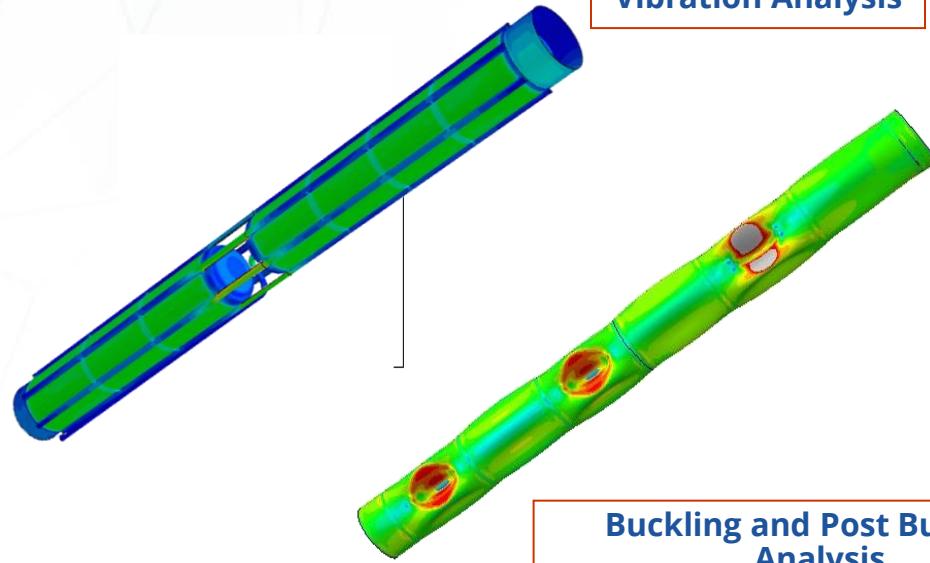
Fairing Separation Dynamic Analysis



Fairing Structural Analysis



Non Linear And Dynamic Analysis of Fuselage



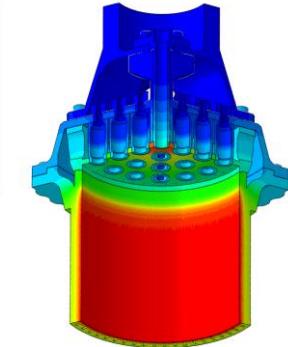
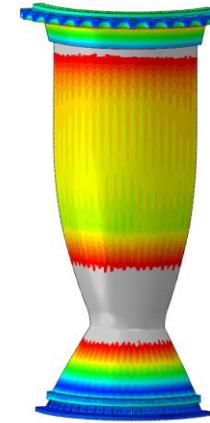
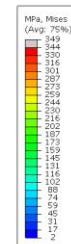
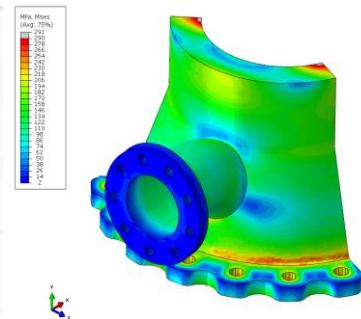
Buckling and Post Buckling Analysis

VE-CTA – Structural Analysis

Today and Future Projects

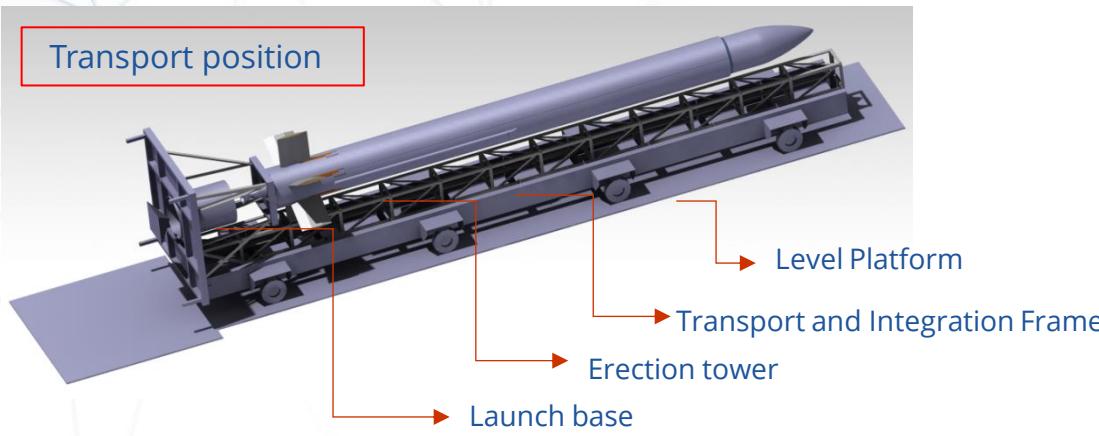
VE-CTA – Rocket Engine Structural Analysis

Engine Structural and Thermal Analysis

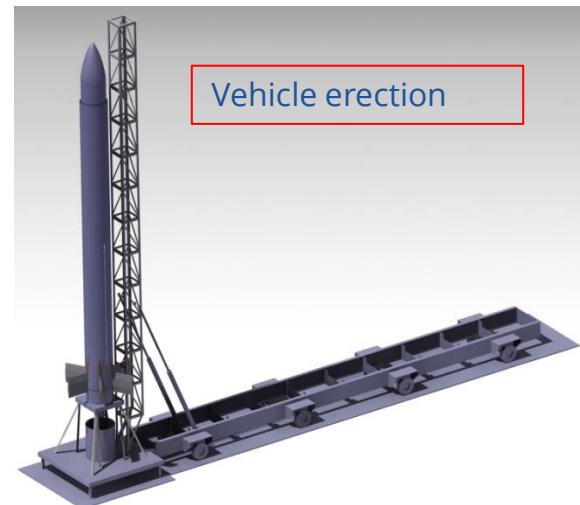


VE-CTA Integration and Launch Systems

Transport position

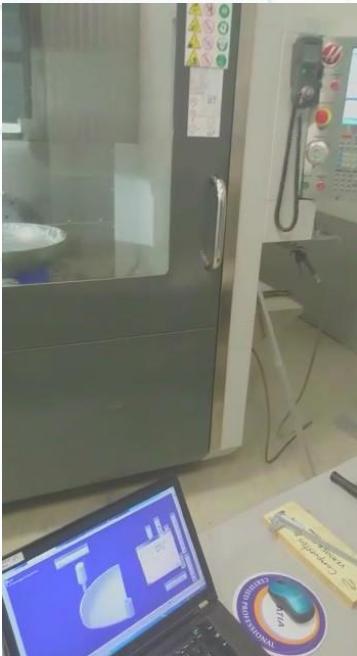


Vehicle erection



Today and Future Projects

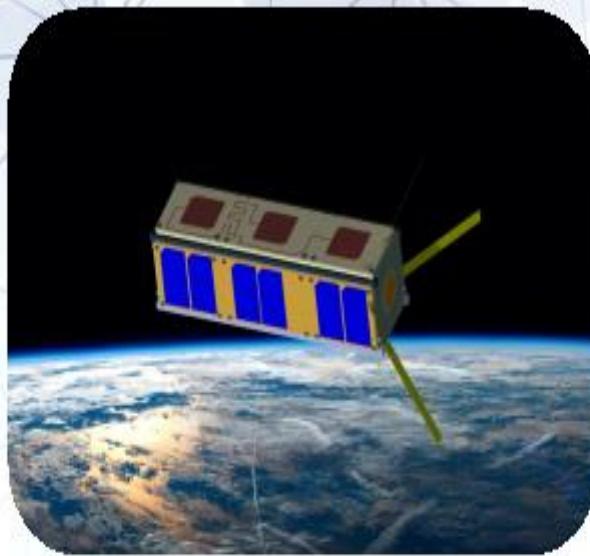
HAAS 5 Axis CNC Machine



VE-CTA Manufacturing

From 3D Virtual Mock Up to Real Manufacturing

Today and Future Projects



University Satellite Programme- Series of CubeSats

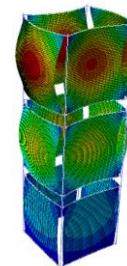
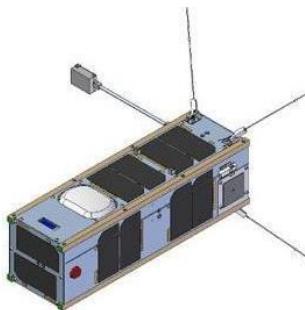
Current Mission:

USAT 1 - GNSS Techniques Technology Demonstrator

- Currently **under development** (Manufacturing and Testing)
- Estimated launch: **Q4 2023**

Future Missions:

1. Quantum Communications Technology Demonstrator
2. ADCS and Navigation for Lunar Missions Demonstrator
3. Missions in collaboration with local and international partners



Lithium Battery Development

CAPABILITIES:

- Design, Integration, Manufacturing and Test of Lithium Batteries for military and Aerospace use.
- Structural and Thermal Design
- Electrical testing
- Environmental Tests (Vibration and Thermo Vacuum)
- Welding Proces



OUR EXPERIENCE: (AEROSPACE USE):

- SAC-C, SAC-D, Satellites
- SAOCOM Satellite (in progress)
- Tronador II Launcher Project
- Design, manufacturing and batteries tunning
- Experimental Vehicles VEx 1^a, VEx 1B and VEx 5A
- Automotive Use battery Manufacturing and design
- (2 Buses for University use among others)

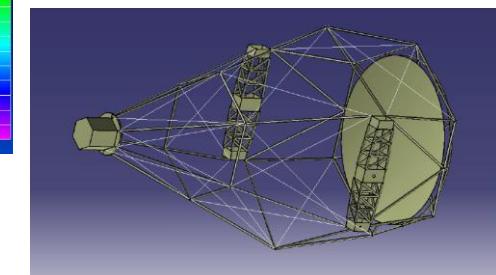
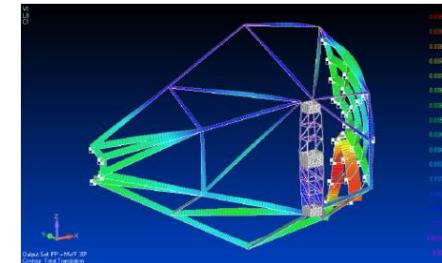


Other Projects and Activities

EU Collaborations:

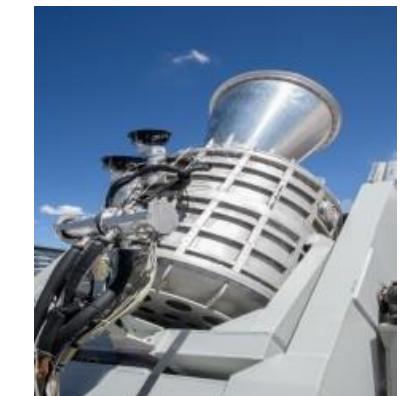
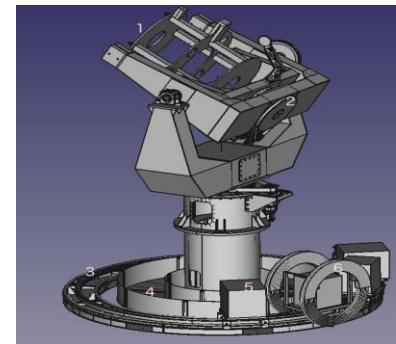
“CTA (Cherenkov Telescope Array)”

Structural and Mechanical Design of 6 meters in diameter of Telescope Mount for the astronomy detection of Gamma Rays



QUBIC EXPERIMENT Collaboration

Structural and Mechanical Design of QUBIC Telescope Mount for the astronomy Deep Space observation

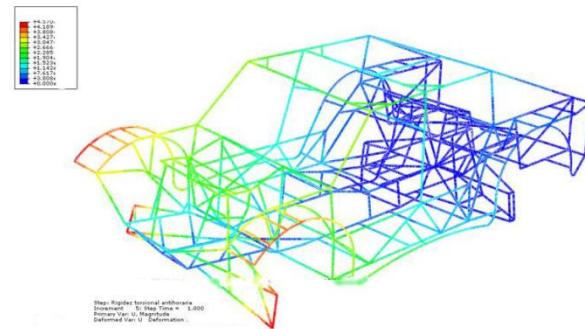


Other Projects and Activities



Argentinian Army Vehicle "GAUCHO"

- ✓ Advice for Structural and Mechanical review
- ✓ Execution of Mechanical Tests for the approval of parts
- ✓ Participation in the Advisory Commission of the Ministries of Defense



ACTIVITIES ON RAILWAY COMPONENTS

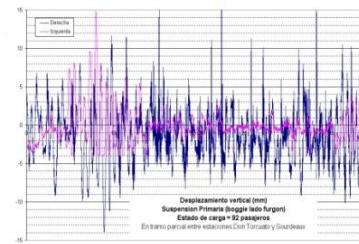
Railway sleepers tests



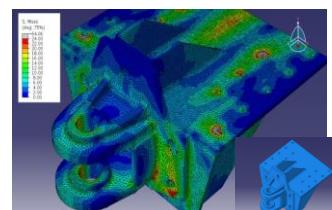
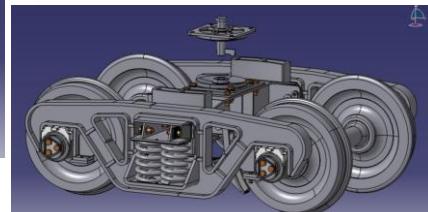
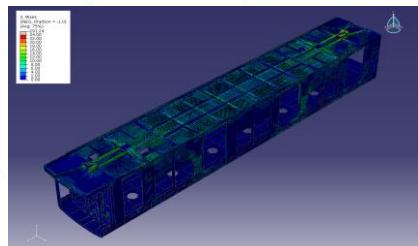
Mechanical tests



Flamability Tests.



Dynamic analysis of structures

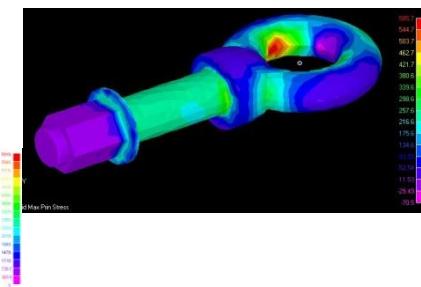
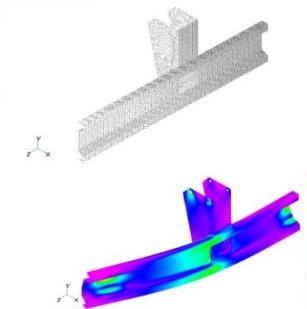


Structural Analysis of train components

ACTIVE AND PASSIVE SAFETY COMPONENTS CERTIFICATION

Evaluation of active and passive vehicle safety systems

- Brake Systems
- Bolts/Eyelets
- Hitching Plates

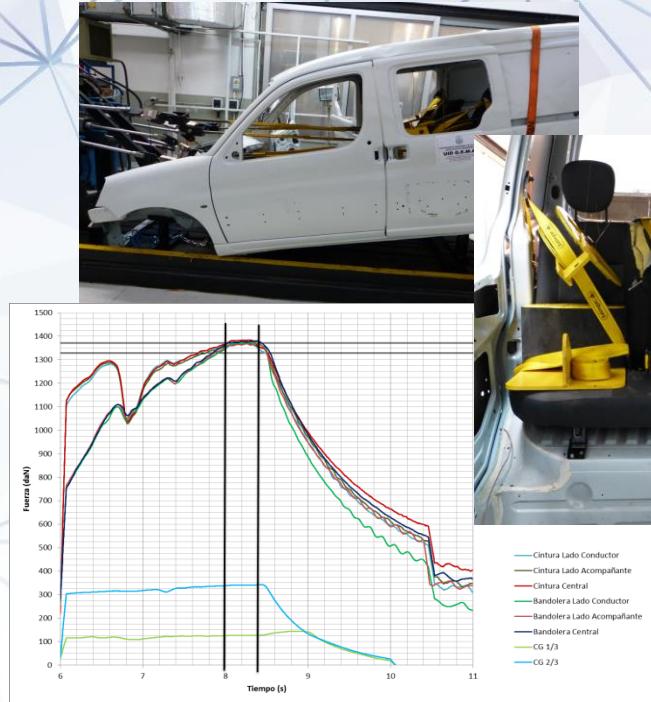


Tests performed under Argentinian and International Laws and Standards:

- Ley 24449 Ley de Transito
- Decreto 779/95 Reglamentario de la Ley de Transito
- Normas IRAM
- Reglamentos de Naciones Unidas (UNECE), entre otras.

ACTIVE AND PASSIVE SAFETY COMPONENTS CERTIFICATION

Testing and Evaluation of seat belts and car seats



Unidad de Investigación y Desarrollo

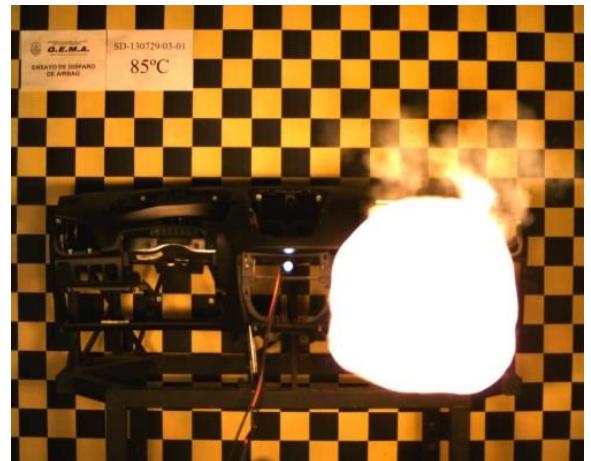
G.E.M.A.

Grupo de Ensayos Mecánicos Aplicados

COMPONENTS TESTING (Safety)

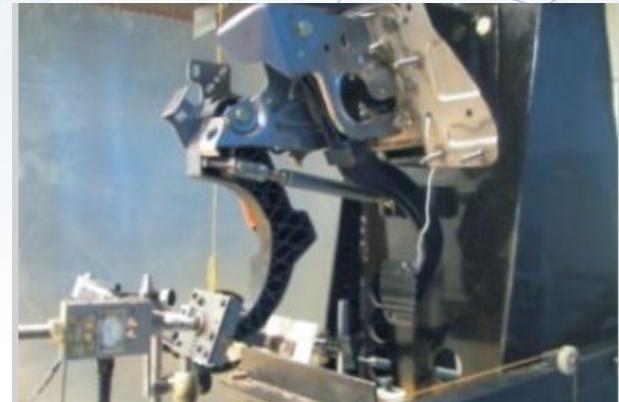
Airbag Testing Laboratory 200 m²

Airbag Firing Tests at Room Temperature, -35°C and +85°C



DIFFERENT COMPONENTS MECHANICAL AND THERMAL TESTING

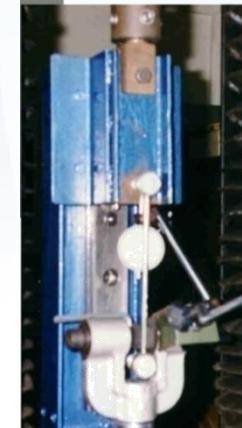
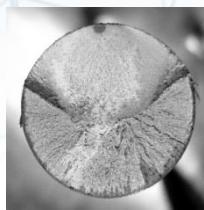
Mechanical tests for components validation



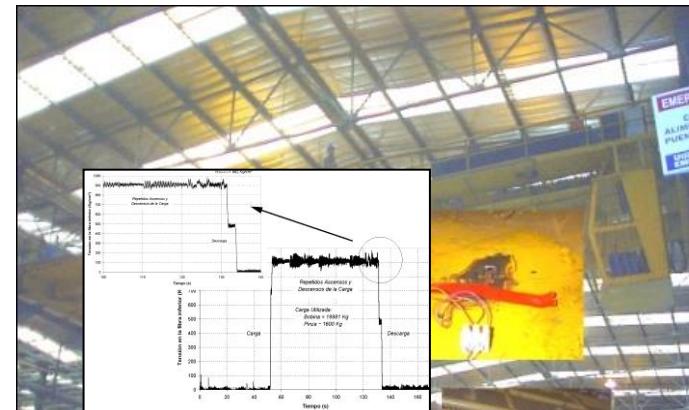
DIFFERENT COMPONENTS MECHANICAL AND THERMAL TESTING

Tests for the determination of Physical Properties on different types of components and materials

- Tensile tests
- Compression Tests
- Flexion Tests
- Cutting Tests
- Hardness Tests
- Torsion tests
- Fatigue tests
- Flammability Tests



Measurements of Accelerations, Displacements and Deformations
(On Site or in Laboratory)



THANK YOU FOR YOUR ATTENTION

CTA (Aerospace Technology Center)

UID G.E.M.A. [Research And Development Unit - Applied Mechanics Testing Group]

UID GFC [Research And Development Unit – Computational Fluid Dynamic Group]

UID LaCLyFA [Research And Development Unit – Boundary Layer Laboratory]



<http://www.gema.ing.unlp.edu.ar>

<http://www.cta.ing.unlp.edu.ar>

La Plata – Buenos Aires - Argentina

